



VG H26-0068 1

#### ADVANCED MIRROR SYSTEM DEMONSTRATOR (AMSD)

# PROGRESS UPDATE AT GOODRICH ELECTRO-OPTICAL SYSTEMS

#### **Enrique Garcia**

Goodrich Electro-Optical Systems 100 Wooster Heights Road Danbury, CT 06810



#### **Agenda**



- Program Objectives and Requirements
- Goodrich Configuration Overview
- Progress Update and Status
  - Facesheet
  - Actuators and Controller
  - Reaction Structure
  - Assembly and Integration
- Test Plan and Program Schedule
- Summary and Conclusions



#### **Agenda**



- Program Objectives and Requirements
- Goodrich Configuration Overview
- Progress Update and Status
  - Facesheet
  - Actuators and Controller
  - Reaction Structure
  - Assembly and Integration
- Test Plan and Program Schedule
- Summary and Conclusions



### **AMSD Program Objectives**



- Diverse government applications require the benefits of:
  - High-payoff large, light-weight mirrors
    - that advance the state of the art, and
    - are rapidly producible, and
    - are affordable
- Specific objectives:
  - Sub-scale demo of the mirror system technology
  - Traceable growth path to deployable, segmented optical systems
  - Provide design features that enable/improve the manufacture, integration, test, and performance of a broad range of operational systems



## Summary of Requirements and Compliance (1 of 2)



STATUS/COMMENT

VG H26-0068 5

#### **REQUIREMENT**

#### Physical

- < 15 kg m<sup>-2</sup> ~ 16.7 kg m<sup>-2</sup> (actuators + CRS repair)

- **Hexagonal shape** Comply (V-notch to eliminate fracture)

- 1.2m to 1.5m point-to-point Comply (1.3 m point-to-point)

#### Mechanical

 Fundamental frequency traceable to full-size flight mirror system Comply

#### Ambient Environment

- **290K to 310K** Comply

- External mechanical Comply

#### Cryogenic Environment

- 30K to 55K Comply

- No mechanical disturbances Comply

#### Survival Environment

- 223K to 353K 223K to 324K (limited by adhesive)

•25K to 353K (cryogenic)

10g quasi static

30g vibroacoustic Comply



## **Summary of Requirements** and Compliance (2 of 2)



VG H26-0068 6

#### **REQUIREMENT**

#### STATUS/COMMENT

Total Surface Error

- 50 nm (rms); 250 nm (P-V) Compliance expected

- Goal 25 nm (rms); 100 nm (P-V) Achievable with additional CCP cycles

Micro-roughness

- 40 Å (rms) Compliance expected

- Goal of 20Å (rms) Compliance expected (< 20Å typical for glass)

- Spatial periods 1 mm to 1 µm Comply

Prescription

- Off-axis parabola Comply

Vertex Radius of Curvature

- 10.000m ± 1mm Comply

Coating

- No coating required Comply (no coating)



#### **Agenda**



- Program Objectives and Requirements
- Goodrich Configuration Overview
- Progress Update and Status
  - Facesheet
  - Actuators and Controller
  - Reaction Structure
  - Assembly and Integration
- Test Plan and Program Schedule
- Summary and Conclusions



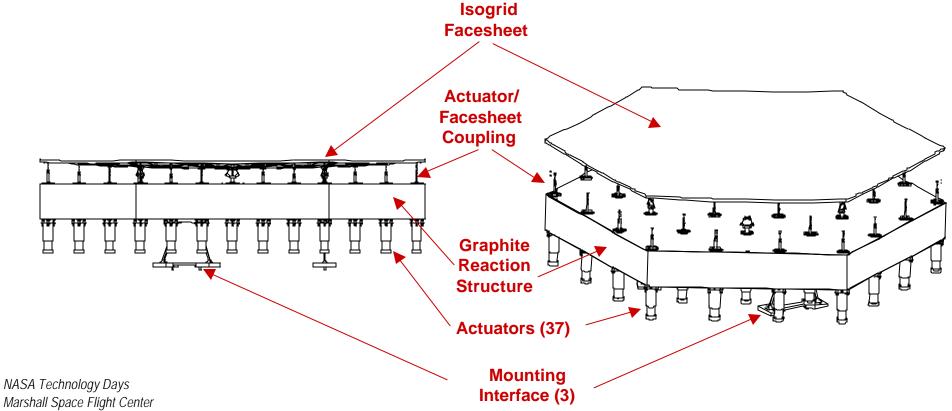
### **Design Concept**



VG H26-0068 8

#### **Architecture:**

- Thin, light-weighted mirror facesheet
- **Array of figure-control actuators**
- Passive, stiff reaction structure
- Mirror facesheet CTE and DL/L matched to reaction structure for cryogenic performance



Marshall Space Flight Center May 22-23, 2002



### **Design Approach**



- Figure-controlled (adaptive) mirror
  - Reduces fabrication/test cost and schedule
  - Maximizes operational system applicability
- Mirror facesheet CTE and DL/L matched to reaction structure
  - Enables cryogenic performance
- Multiple material options (mirror/reaction structure pairings) from a single architecture
- Design optimization to mission constraints
  - Environment Performance
  - Cost Schedule
  - Actuator design is common
  - Mirror facesheet and reaction structure details are material dependent
  - Traceability assessments are material dependent
    - Influenced by specific application
    - Material dependent processes and facilities



### **Design Approach Benefits**



- Cost and schedule effective optical manufacturing
  - Non-recurring investment in tooling, followed by rapid fabrication of multiple matched facesheets
  - Work with large tools for majority of processing time
  - Actuators for low spatial frequency correction
  - Computer Controlled Polishing (CCP) for remainder
- Cost and schedule effective system-level operations
  - Actuators provide radius and figure adjustment at test and operating temperatures
  - Flexible substrate allows increased shape correction
  - Reduced reliance on ground testing



## **Approach for Optical Fabrication: Stressed Mirror Polishing Overview**



- Basic fabrication process referred to as "Stressed Mirror Polishing" (SMP)
  - R2 (backside) of optic is fabricated to a sphere
  - Optic is held, by vacuum, against aspheric blocking body
  - Blocking body asphericity is negative of desired R1 asphericity
  - R1 is polished to a sphere
  - When vacuum is released, R1 will "spring" to desired asphere
  - Touch-up polishing after mounting to actuators
- SMP process has several advantages
  - Spherical fabrication process is fast, smooth, and simple
  - Minimal optic handling reduces risk most of the time the optic is mounted to granite blocking body
  - For multiple optics of same form, investment in blocking body paid off early

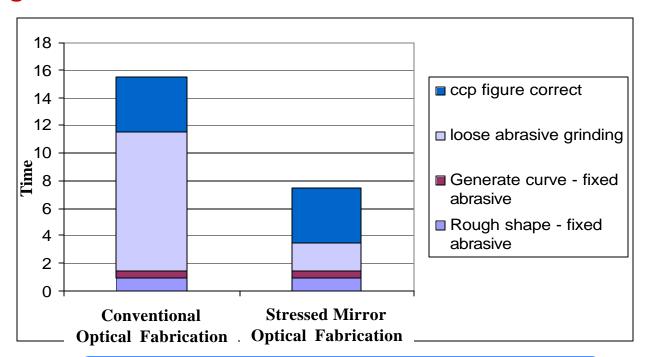


#### Fabrication of Fast Aspheres by SMP

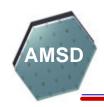


VG H26-0068 12

- Taking advantage of thin section, bend the mirror to look like a sphere; grind and polish a sphere
- Large tools can remove material damage layer much faster
- Return to small-tool processing of unstressed asphere for final figure correction



Recurring and non-recurring efforts tailored based on quantity of units.



#### AMSD Approach Leverages Demonstrated Large Optical Systems Technology



VG H26-0068 13

#### HALO

- 3-meter diameter Primary Mirror Assembly
- 30 kg/m<sup>2</sup>
- tested at 100 Kelvin

#### LAMP

- 4-meter diameter Primary Mirror Assembly
- room temperature High Energy Laser System

#### ALOT

- 4-meter diameter lightweight telescope for space operation
- 70 kg/m² PMA
- room temperature imaging system

#### LOS

two four-meter diameter segments of 11-meter, f/1.25 primary mirror

Large, segmented mirrors and telescopes benefiting from shape-controlled technology are demonstrated.



#### **Agenda**



- Program Objectives and Requirements
- Goodrich Configuration Overview
- Progress Update and Status
  - Facesheet
  - Actuators and Controller
  - Reaction Structure
  - Assembly and Integration
- Test Plan and Program Schedule
- Summary and Conclusions



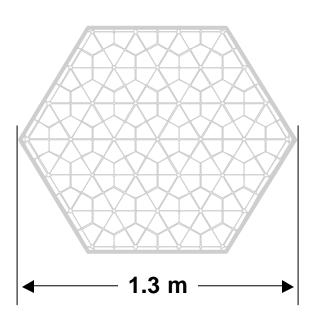
### **Progress Update and Status**



VG H26-0068 15

## **Light-Weighted Thin Facesheet:**

- Facesheet design
- Optical fabrication (SMP)
- Light-weighting and edging
- Fracture and recovery
- Status



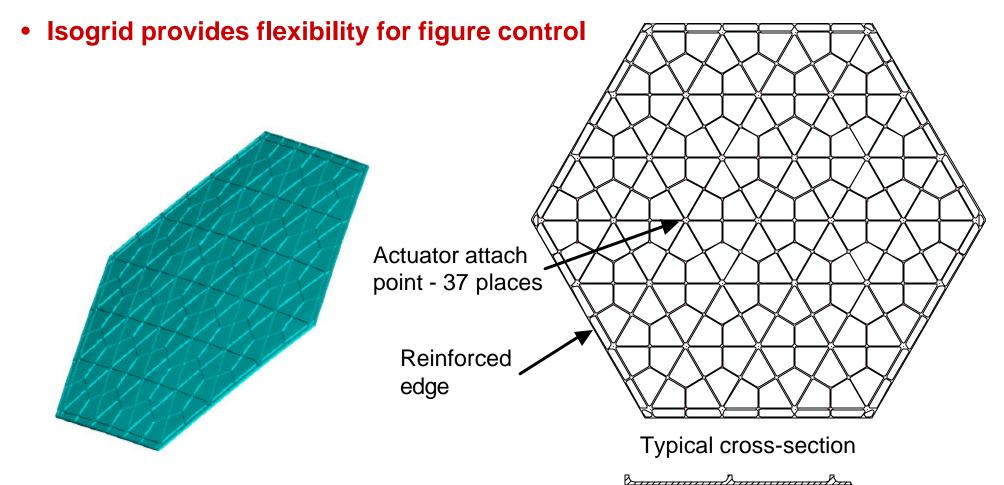


#### **Facesheet Design**



VG H26-0068 16

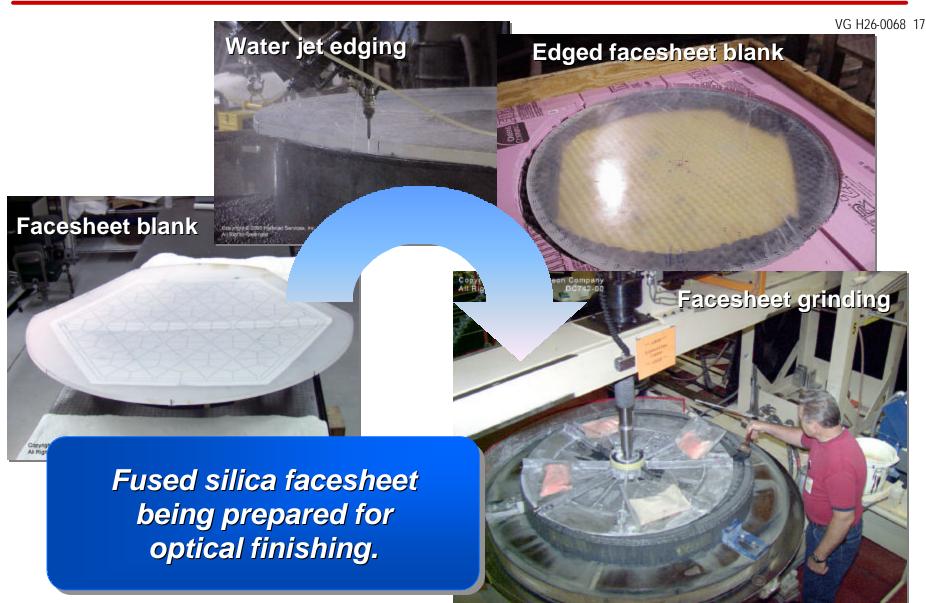
- Material is fused silica for CTE homogeneity (but could be ULE)
- Isogrid provides stiffness for 1-G support





## Optical Fabrication: Preparing the Blank for Optical Finishing



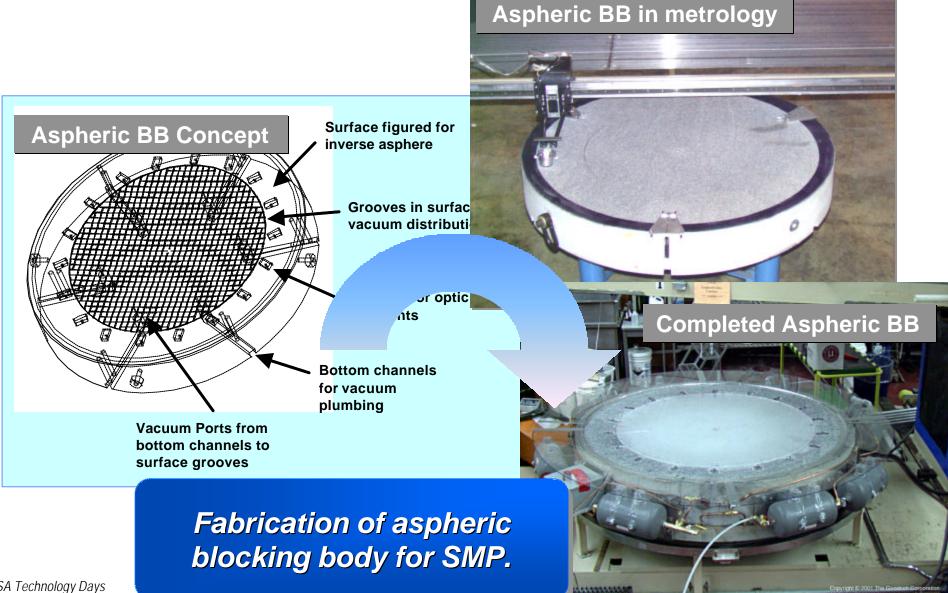




## Optical Fabrication: Preparing the Aspheric blocking Body



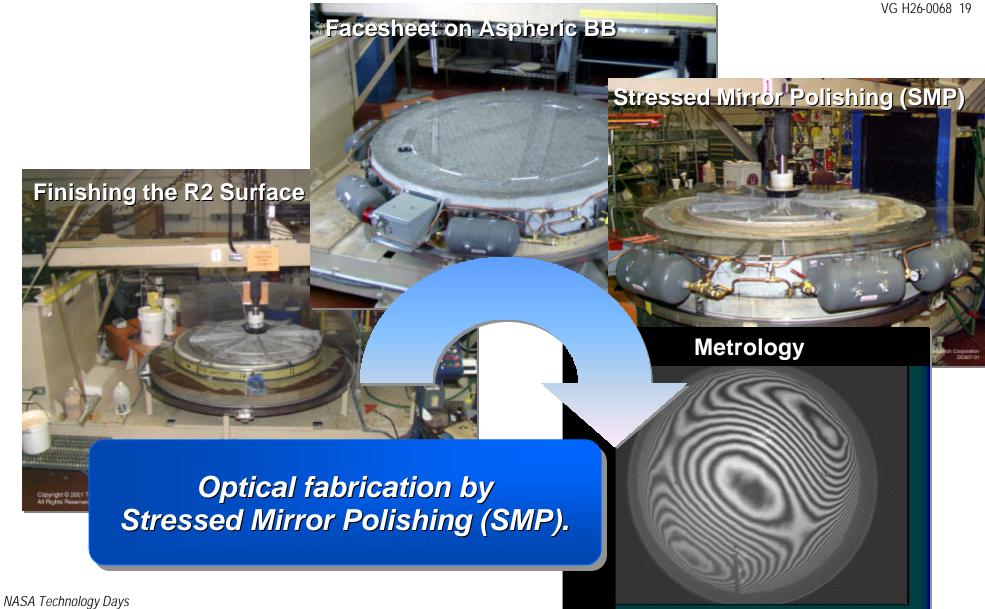
6-0068 18





## **Optical Fabrication:**Optical Finishing by SMP

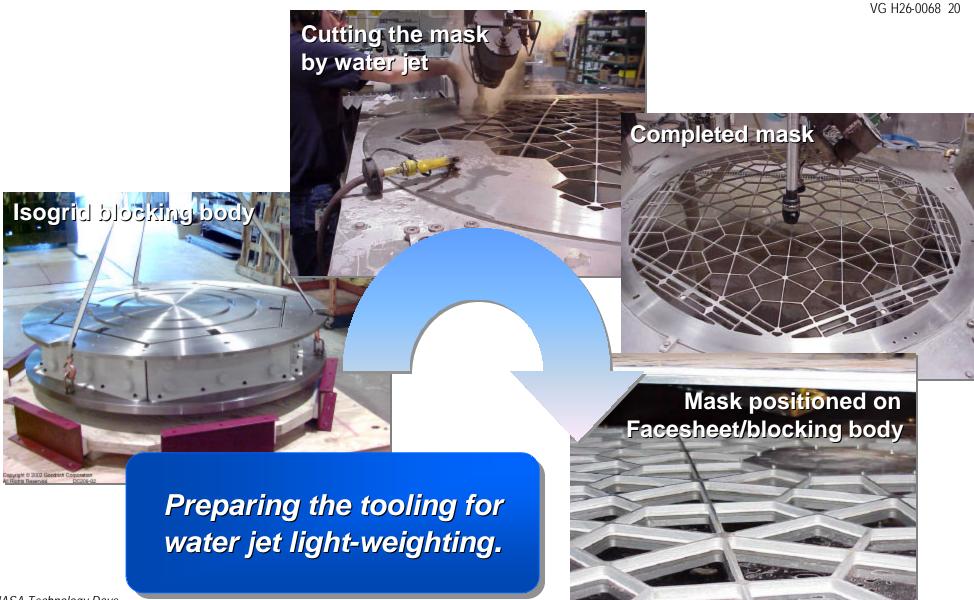






## **Optical Fabrication: Tooling for Water Jet Milling**

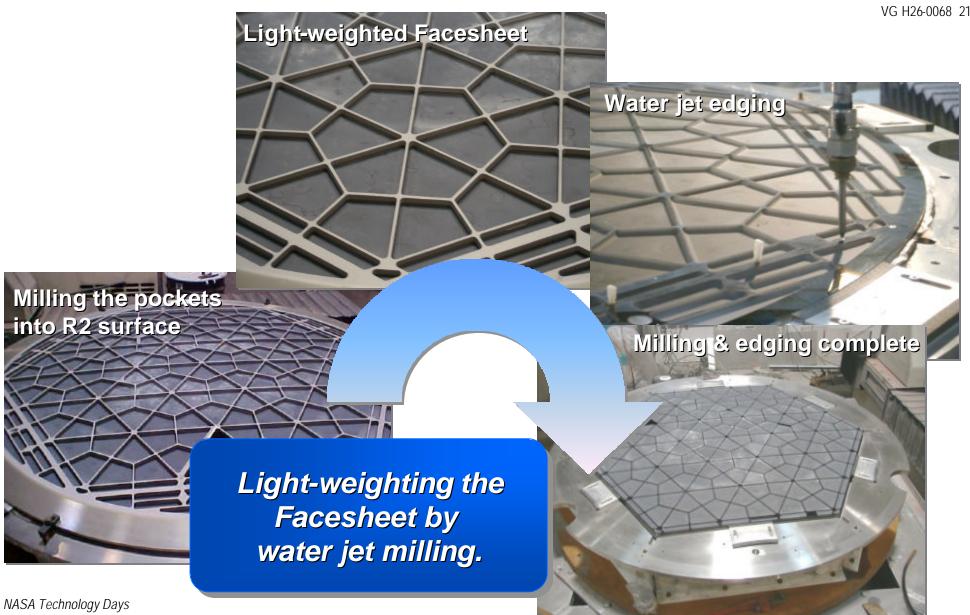






## **Optical Fabrication: Light-Weighting and Edging**





Marshall Space Flight Center May 22-23, 2002



## **Optical Fabrication:** Fracture and Recovery



VG H26-0068 22



Fracture during routine grinding of R2

Optical fabrication completed as planned; no loss of actuators.



#### **Facesheet Status**



VG H26-0068 23

### Facesheet Progress and Status:

- Fracture "repair" complete
- Optical fabrication by SMP complete
- Light-weighting by water jet milling complete
- Edging by water jet complete
- Beveling and stress relief in progress
- Preparations for Facesheet Subassembly underway

Optical fabrication complete; Stressed Mirror Polishing advantages demonstrated.



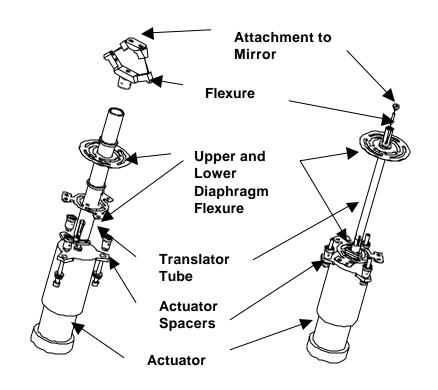
## **Progress Update and Status**



VG H26-0068 24

#### **Actuators and Controller:**

- Design overview
- Controller
- Test results
- Status



**Bipod Assembly** 

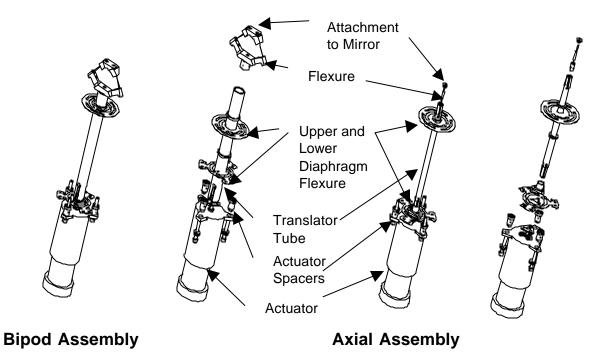
**Axial Assembly** 



## **Actuation System Design**



- Actuator assemblies (6 "Bipod", 31 "Axial") react all loads
- Similar to HALO (also cryogenic)
- Upper and lower diaphragm flexures stabilize translator & flexure, provide shear and moment load path for bipods
- Allows easy access to actuators for installation and servicing





NASA Technology Days Marshall Space Flight Center May 22-23, 2002



## **Actuator Design Parameters Summary**



Performance Parameter	Reason	
Stroke	Ground test and post- deployment capture range	
Resolution	Performance (quantization)	
Mass	Observatory flow-down	
Reliability and Lifetime	Maintain low performance risk	
Room and Cryogenic Performance	Operational performance and efficient ground-test strategy	
Low (zero) Power Dissipation	Thermal maintenance	
	Operational efficiency	
Axial Stiffness	PMA Dynamics	
Compatibility with Cryo-Appropriate Command/Power Structure	Minimize wire count for deployment  Maintain reliability and redundancy	

- Goodrich has selected stepper-motor based actuator from Moog-Shaeffer Magnetics Division (SMD):
  - Derived from NASA-funded cryo actuator studies
  - Engineered under Goodrich and Moog IRAD



## Actuator Design: Space Rated Materials for Cryogenic Operation



VG H26-0068 27

		VG H2
	Titanium Alloy (6AL-4V)	General Construction: Housing/Transducer, Motor Housing, Thrust Rod, Spring/Nut Element, Output Flange
	Stainless Steel (440C)	Bearings, Harmonic Drive Wave Generator
CM 5 10	Stainless Steel (Nitronic40)	Output Leadscrew
INCH 2 3 4	Stainless Steel (15-5PH)	Harmonic Drive Circular Spline and Flex Spline
	Stainless Steel (416)	Motor Rotor

- Actuator internal bearings, harmonic drive wave generator, lead screw/nut use dry lubricant for low room temperature friction and excellent molecular bonding
- Materials are selected specifically for cryogenic application and compatibility
- Structural members are sized to perform over life with ample margin
- Fasteners are generally titanium with only a few 416 stainless steel (thermally matched)

Performance demonstrated at 30K, consistent with AMSD requirements.

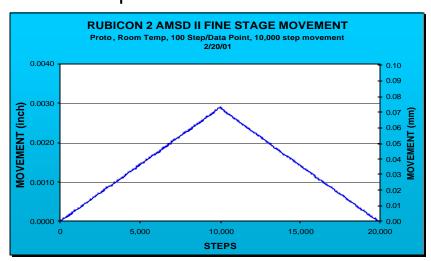


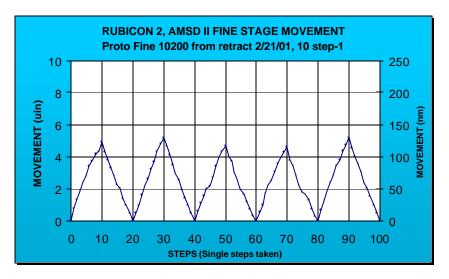
## Actuator Design Verified at RT and Cryo



VG H26-0068 28

#### Room temperature tests:





#### **Example Data:**



- Average Step Size = 7.3 nanometers
- Standard Deviation = 3.5 nanometers
- Meets requirement for max. step size < 20 nanometers</li>

#### RT vs. Cryo Performance:

- Same general behavior when unloaded
- Some units have exhibited anomalous behavior under load at cryo
- More extensive cryo tests underway



## **Actuator Electronics:**Design Requirements



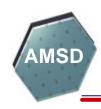
- Architecture must be traceable to deployable, space-borne applications
- Minimize wire-count between nodes and central controller
  - Hold actuator position with minimal actuator power
  - Minimize overall power dissipation
- Architecture must be scaleable to large arrays of actuators
- Circuit design must be traceable to operation at 30K
  - Operate over temperature range from 30 to 293K
  - Active devices MOS (bipolar processes freeze-out at T < ~70K)</li>
- AMSD Electronics design does not limit bandwidth (» 0.1 Hz update rate)



#### **Actuator Electronics:** Architecture



- Control Computer (CC) and array of actuator nodes form "ring"
  - Serial communication minimizes wire count
  - Resistive isolation between nodes prevents fault propagation
  - Easily provides required throughput
  - Ultimate bandwidth limit is motor dynamics
- All system "intelligence" in software executing on CC
  - Directly commands motor windings to desired states
  - Operationally flexible, upgradeable
- "Dumb" hardware at nodes
  - Translate winding state commands to winding drive currents
  - Minimizes hardware complexity, maximizes system flexibility
- Topology is inherently scaleable
  - Change CC software required
- AMSD drive electronics are "warm" (located outside vacuum chamber)

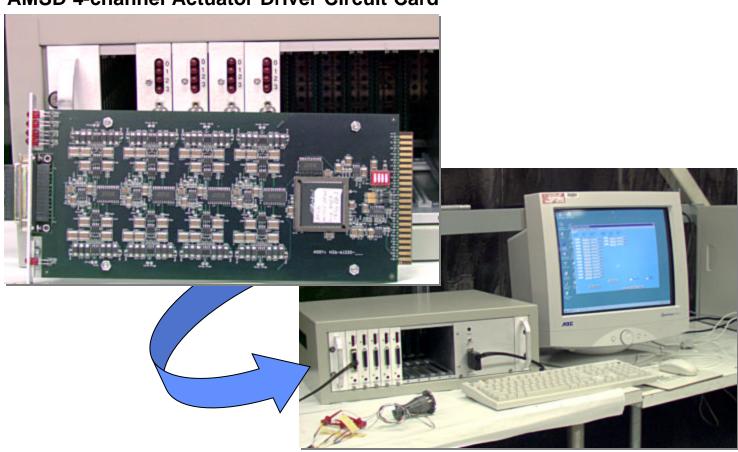


### **Actuator Electronics/Controller**



VG H26-0068 31

#### **AMSD 4-channel Actuator Driver Circuit Card**



Drive electronics, controller, software and cabling are complete.



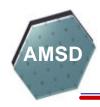
#### **Actuators/Controller: Status**



VG H26-0068 32

- Actuators/Controller Progress and Status:
  - All 37 AMSD units have been delivered
  - Drive electronics and controller complete
  - Performance tests at RT complete
  - More extensive cryo testing under load underway
  - Integration of actuation components with Reaction Structure underway

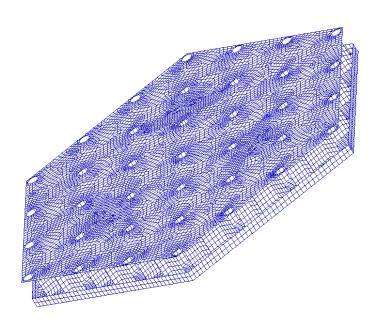
All actuators delivered; basic design/performance verified.



## **Progress Update and Status**



- Composite Reaction Structure:
  - Design and construction
  - Modeling
  - Delamination and repair
  - Cryo test results
  - Status





### **CRS: Design and Construction**



- Graphite cyanate-ester for CTE match with glass Facesheet
- Mounts actuators and flexures; reacts loads from masses and figuring
- Interfaces to external mount
- Designed and manufactured by ATK

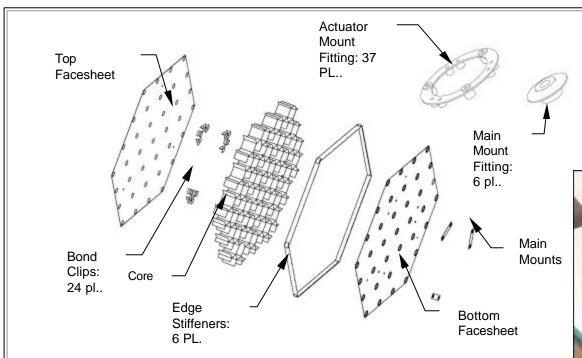


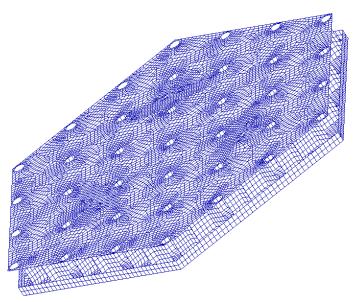
Photo of completed unit

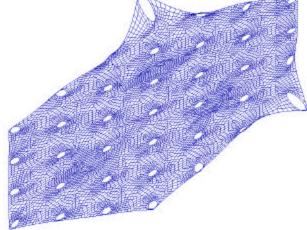




## **CRS: Modeling and Analysis**







- NASTRAN Detailed Finite Element Model
  - Layered shell elements
  - 37 concentrated mass elements representing actuators and INVAR fittings
  - Simple support at main mount locations
- Predicted response of baseline structure meets requirement (1st mode reduced to due to repair)

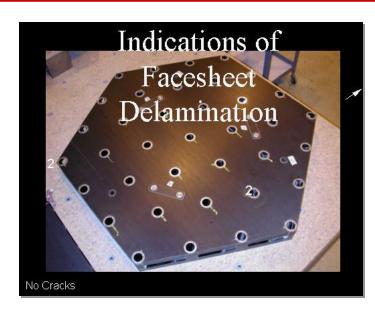




## **CRS: Delamination and Repair**



- Post cryo cycling inspection revealed voids in facesheets:
  - Voids at specific orientation over core ribs
  - Attributed to alignment of facesheet plies relative to core cells
- Repaired by composite straps between top and bottom facesheets:
  - At every location of actual or likely delamination
  - Repair complete
  - CRS cryo cycled at the XRCF
- Origin of failure and solutions identified



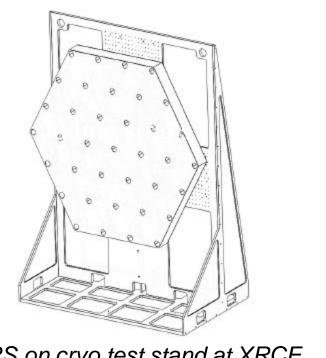




#### **CRS: Tests and Results**



- **Tests conducted on repaired CRS:** 
  - Thermal cycle to 25 K at XRCF
    - No metrology for distortion
    - No further delamination.
  - 10-G static load test at ATK
  - Thermal characterization at XRCF
    - Instrumented for distortion measurements
    - Two cycles to 25 K
    - Measured distortion within acceptable limits



CRS on cryo test stand at XRCF



**CRS: Status** 



VG H26-0068 38

- CRS Progress and Status:
  - Fabrication complete
  - Delamination:
    - Causes and solutions identified
    - Repair complete
  - Cryo test of repaired unit complete
  - Integration with actuation system underway

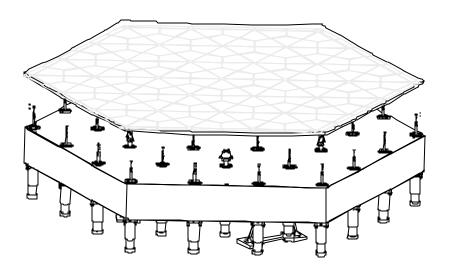
CRS complete and verified by analysis and measurement.



# **Progress Update and Status**



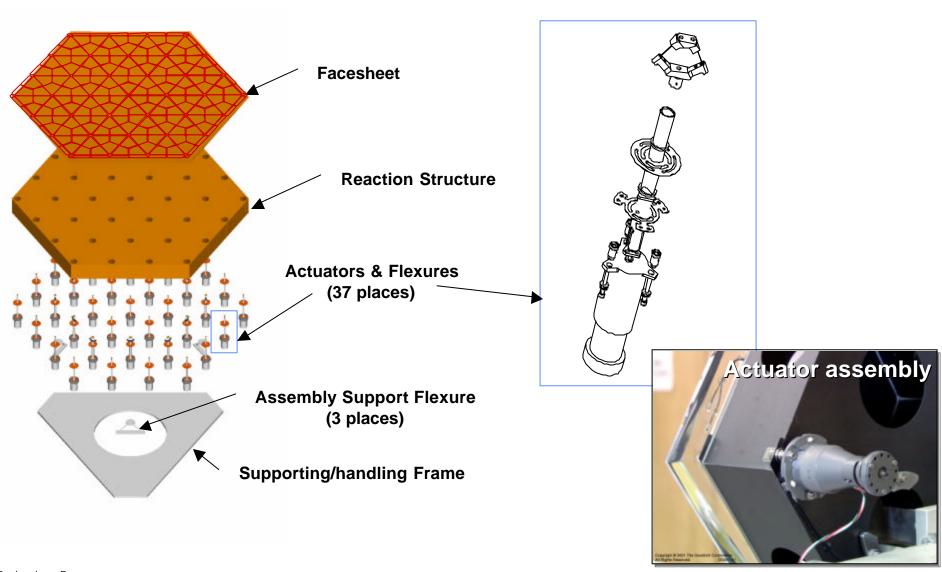
- Assembly and Integration:
  - Assembly description
  - Integration process
  - Status and Plans





# **Assembly Description**

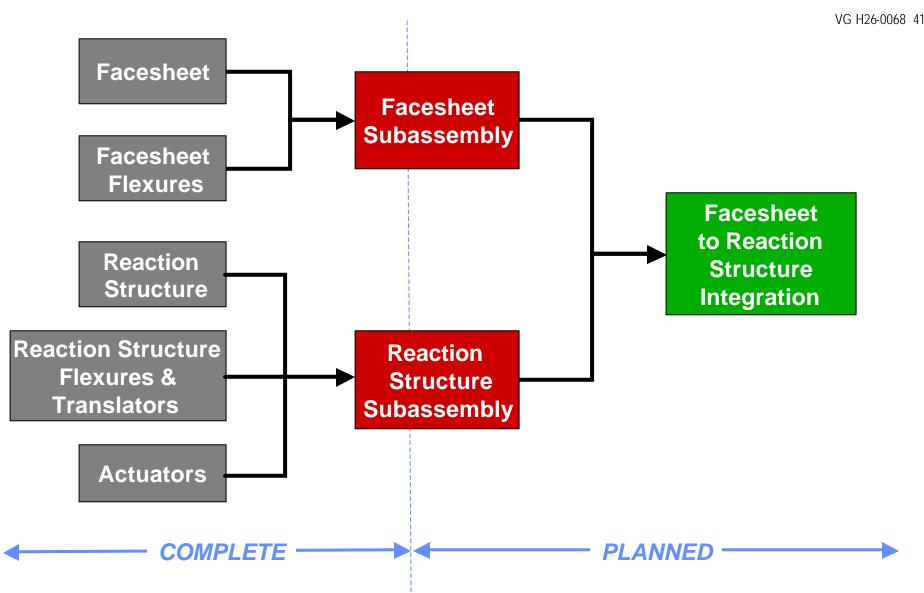


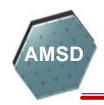




# **Hardware Integration Flow**







## **Assembly and Integration: Status**



VG H26-0068 42

- Assembly and Integration Progress and Status:
  - Design and process definition complete
  - Tooling design and fabrication partly complete and continuing
  - Reaction Structure/Actuator subassembly in progress
  - Preparations for Facesheet Subassembly underway

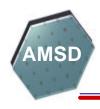
Assembly and Integration in progress per plan.



#### **Agenda**

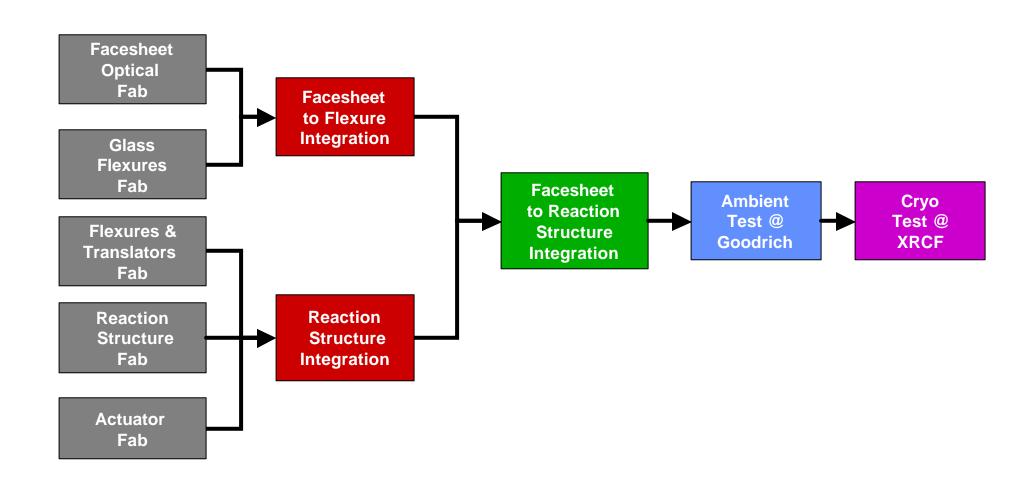


- Program Objectives and Requirements
- Goodrich Configuration Overview
- Progress Update and Status
  - Facesheet
  - Actuators and Controller
  - Reaction Structure
  - Assembly and Integration
- Test Plan and Program Schedule
- Summary and Conclusions



# **Process Flow Summary**



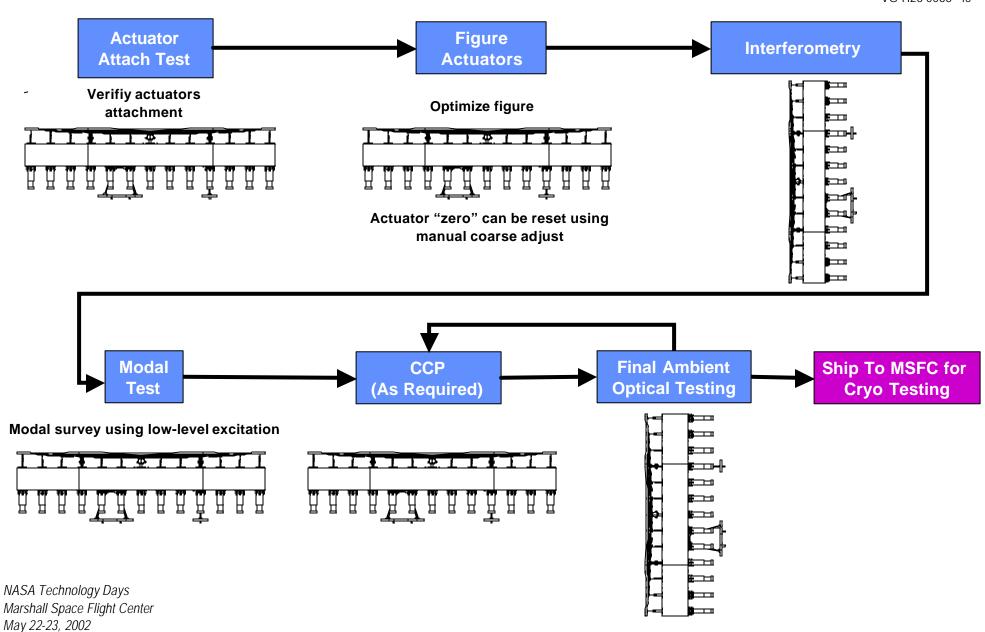




#### **Test & Verification**

(RT at Goodrich)

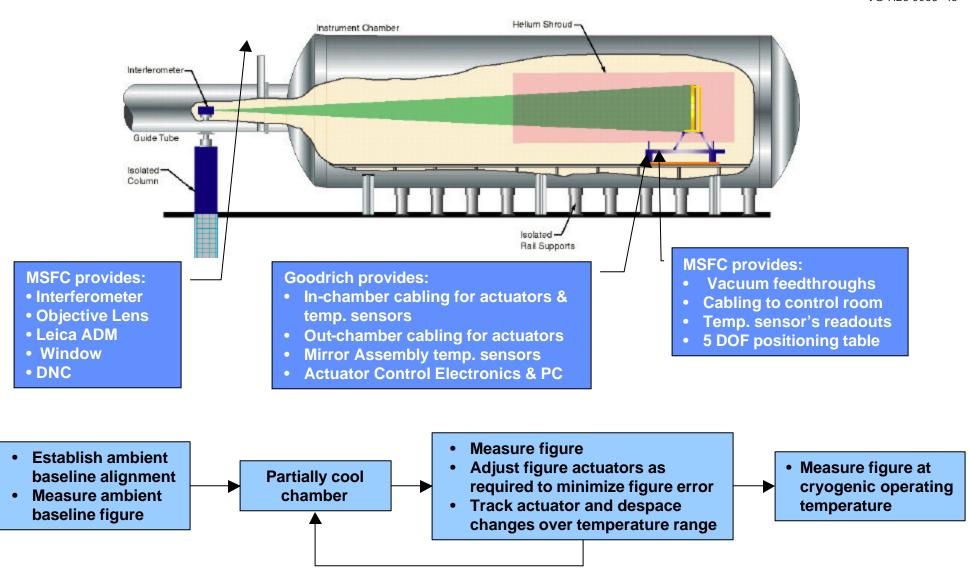






# **Cryo Test Arrangements at the XRCF**



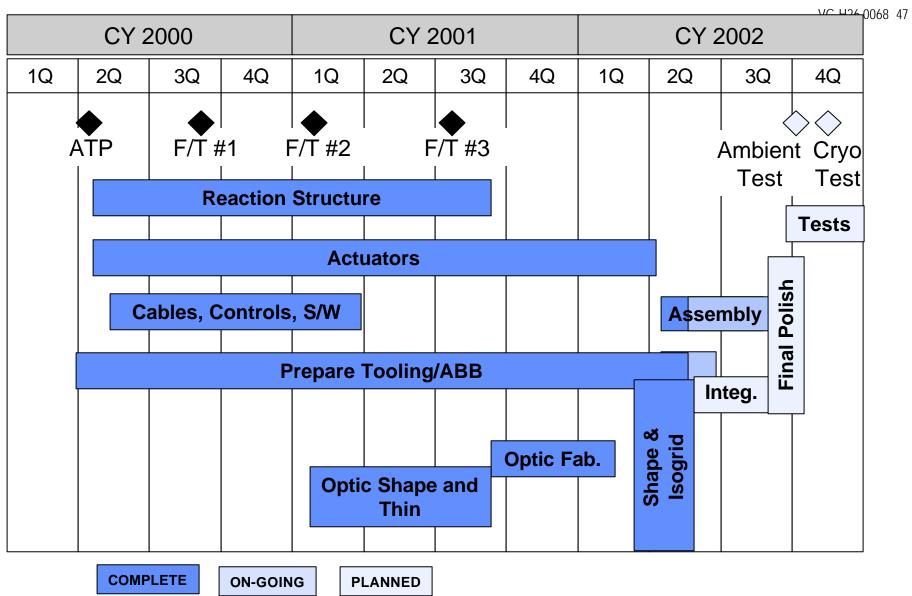




#### **Top-Level AMSD Program Schedule**



(as of 5/22/02)

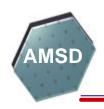




#### **Agenda**



- Program Objectives and Requirements
- Goodrich Configuration Overview
- Progress Update and Status
  - Facesheet
  - Actuators and Controller
  - Reaction Structure
  - Assembly and Integration
- Test Plan and Program Schedule
- Summary and Conclusions



# **Design Traceability Overview**

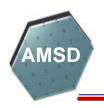


VG H26-0068 49

- Our AMSD design is traceable to operational systems:
  - Flexible facesheet enables efficient figuring of a readily produced substrate
  - 'Displacement' type actuators
     stiffen facesheet against reaction
     structure while providing shape
     control
  - Reaction structure utilizes a high stiffness-to-mass material that is amenable to efficient structural forms

- Material choices can be tailored to specific applications (facesheet and/or reaction structure)
- 'External' actuator permits adoption of improved designs
- Mass and stiffness changes are addressed without disruption to key facesheet/actuator design and manufacturing details

Our AMSD design is fully traceable against the SOW requirements.



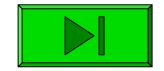
## **Summary**



VG H26-0068 50

- Goodrich's AMSD architecture provides robust accommodation for a broad range of system and mission requirements:
  - Readily accommodates alternative petal geometries
  - Readily accommodates alternative materials
  - Provides opportunity to trade mass, stiffness, and segment size for optimal mission responsiveness
  - Manufacturing technique is cost/schedule effective for multiple builds

Rapid optical fabrication and isogridding have validated the recurring benefits of the AMSD manufacturing processes.







VG H26-0068 51

#### ADVANCED MIRROR SYSTEM DEMONSTRATOR (AMSD)

#### AMBIENT AND CRYOGENIC TEST PLANS AT GOODRICH ELECTRO-OPTICAL SYSTEMS

#### Mark Furber

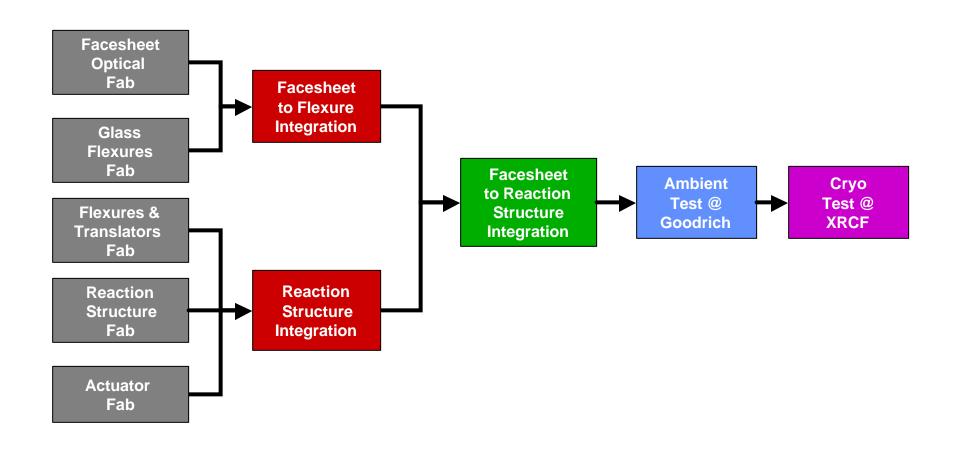
Goodrich Electro-Optical Systems 100 Wooster Heights Road Danbury, CT 06810



## **Process Flow Summary**



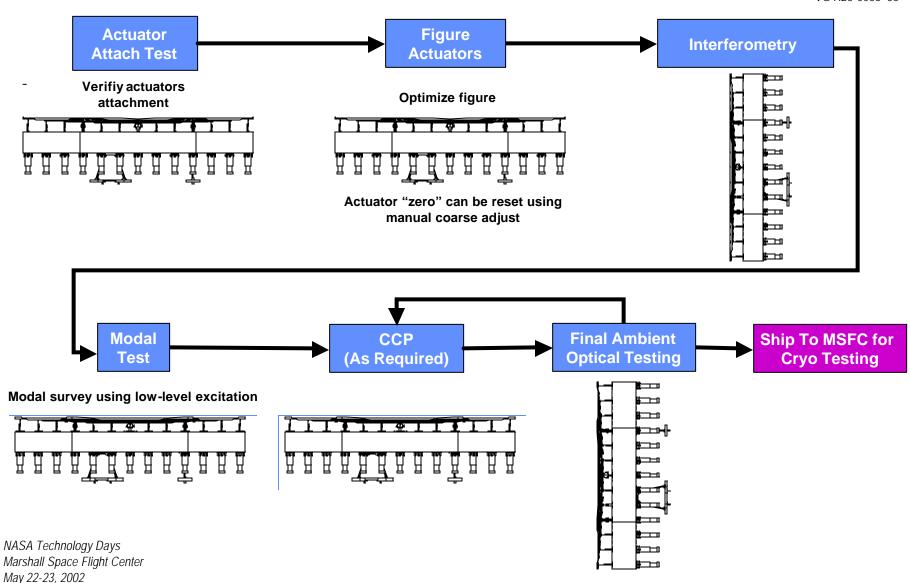
VG H26-0068 52





# Test & Verification (RT at Goodrich)



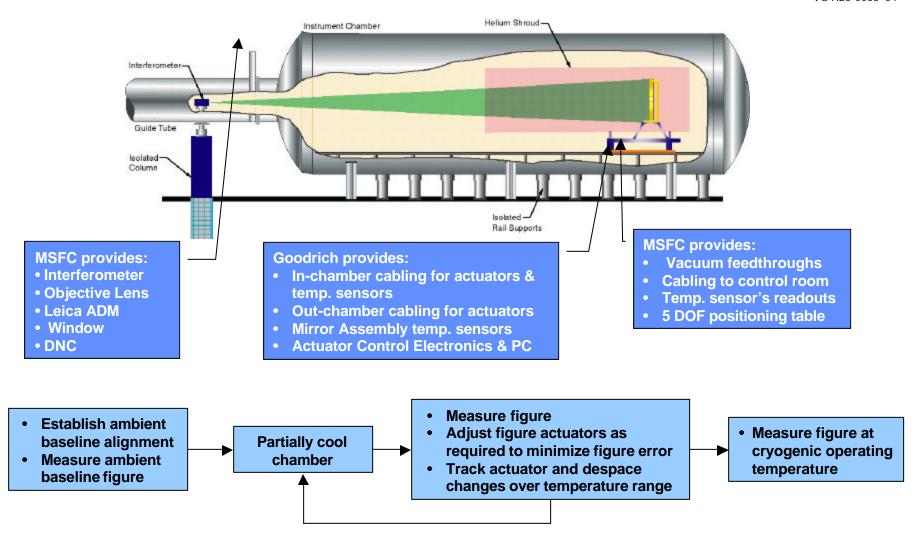




#### **Cryo Test Arrangements at the XRCF**



VG H26-0068 54





#### Test & Verification Plan (Summary)



VG H26-0068 55

#### Ambient at Goodrich

- Install and align test article
- Verify software and electronic functionality
- Manual figure correction
  - Visual
- Determine noise levels (OPD, influence functions)
- Influence function measurement
- Optimize figure
  - Use measured influence functions
- Initial data reduction
  - Document results
- Pack and ship to MSFC

#### Ambient and Cryo at MSFC

- Install and align test article
- Check out interferometer and software and electrical Interfaces
- Room temperature characterization as done at Goodrich
- Radius of curvature monitoring throughout
- Cryo Test
  - Cool down, adjusting figure intermittently as needed
  - At cryo:
    - Check actuator functionality
    - Manually optimize figure
    - Measure influence functions & associated noise
    - Optimize figure
    - Actuator/influence function characterization measurements
  - Return to RT, adjusting figure intermittently as needed
- Remove test article



### **Top-Level AMSD Program Schedule**



(as of 5/22/02)

